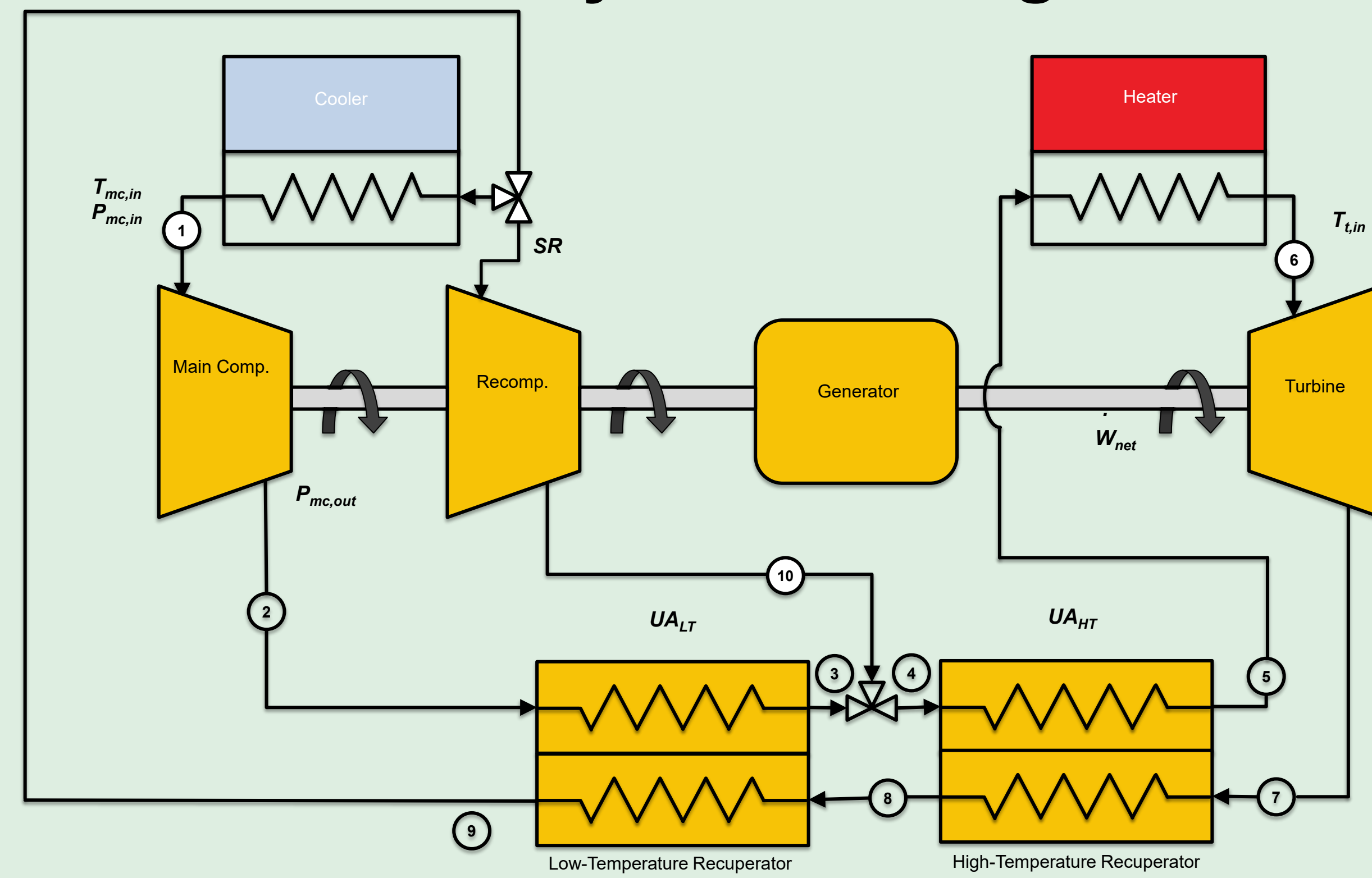


Supercritical Carbon Dioxide Power Cycle Modeling and Additive Manufactured Turbine Experimentation

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This work presents an integrated study of supercritical carbon dioxide power cycles, combining cycle-level modeling, turbomachinery loss prediction, and preparation for experimental validation. A detailed modeling framework incorporating empirical aerodynamic, windage, and leak-age loss correlations enables accurate system-level performance predictions across a range of power scales without relying on computationally intense methods. Multi-objective optimization and parametric studies reveal trade-offs between cycle efficiency and system compactness, showing how turbomachinery losses influence small-scale system performance and push optimal compressor inlet pressures below the critical point. Complementing the modeling, a test facility was designed and constructed to be capable of 800°C and 11 MPa operation, including management of complex auxiliary cooling flows. This facility was validated and will be used to test a 30 kW turbine-generator system additively manufactured from Haynes 282 to characterize and validate loss modeling predictions. This integrated approach provides pathways toward higher-efficiency, scalable cycles.

Cycle Modeling



Integrated Loss Models

- Heat exchanger performance: discretized effectiveness-NTU model [3], [4]
- Heat exchanger sizing: analytical/empirical model [2] based on Blasius [5] and Dittus-Boelter [6] correlations
- Turbomachine aerodynamic losses: specific speed and specific diameter curves compiled by Balje [7] and digitized by Sondelski [8]
- Radial turbomachine leakage: Curve fit function of geometric clearances proposed by Qi [9] based on data from Futral and Holeski [10]
- Radial turbomachine windage: analytical model given by Vilim [11] with coefficients from Daily and Nece [12]
- Labyrinth seal leakage: empirical model developed by Egli [13] and Aungier [14],[15], and modified by Grotjan [2]
- Generator windage: empirical model developed by Aungier [14],[15]

See associated paper for list of references.

Experimentation

Figure 1: Schematic of RCBC with model inputs. Gold components integrate more sophisticated loss modeling than typical RCBC models in the literature.

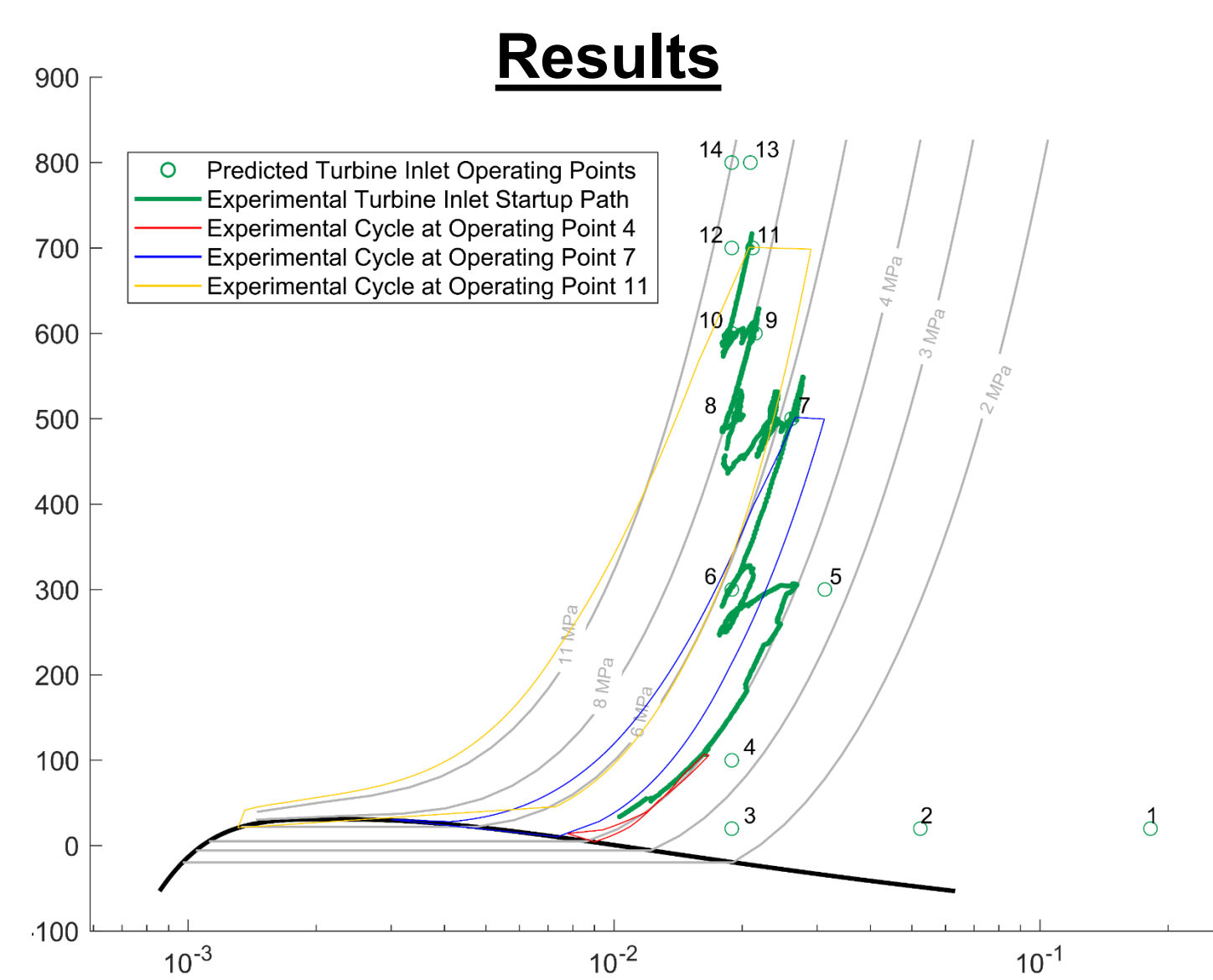


Figure 8: T-v diagram showing the proposed turbine-inlet start-up path, experimental data obtained using an orifice in place of the turbine, and full-cycle loops at operating points 4, 7, and 11.

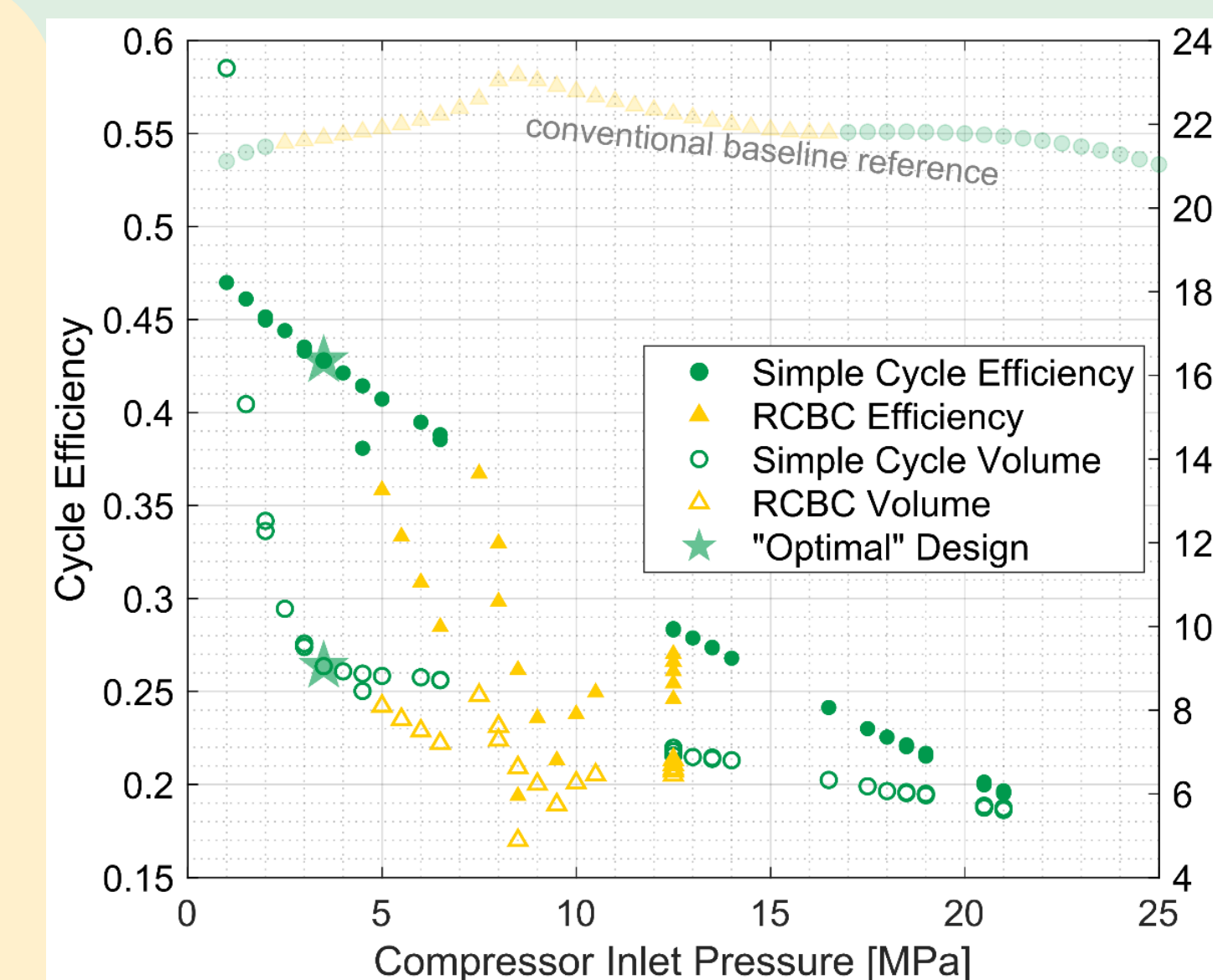


Figure 2: 500 kW power scale: Efficiency and normalized volume with loss models. Conventional baseline efficiency for constant turbomachine efficiency model is overlaid.

Results

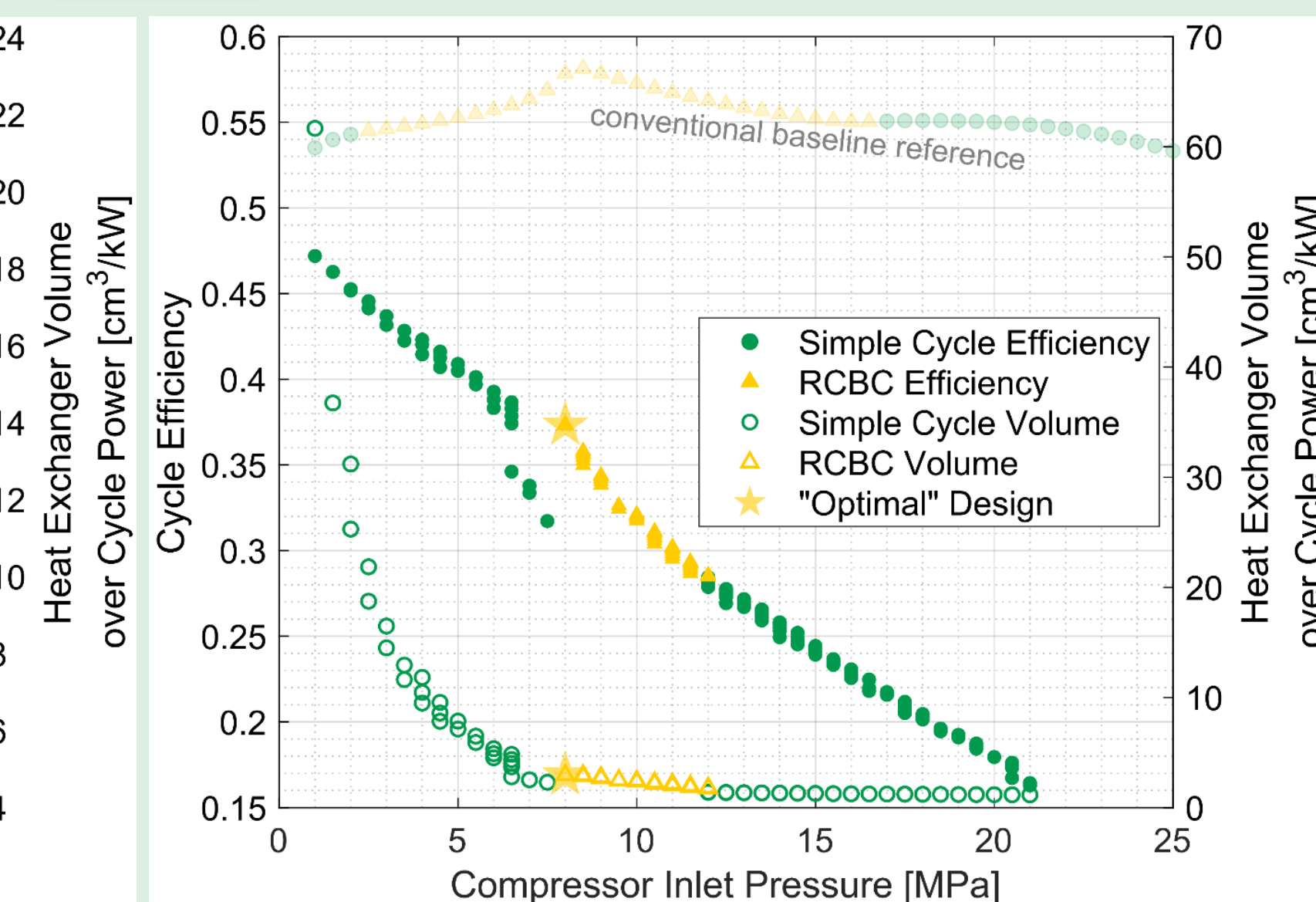


Figure 3: 50 kW power scale: Efficiency and normalized volume with loss models. Conventional baseline efficiency for constant turbomachine efficiency model is overlaid.

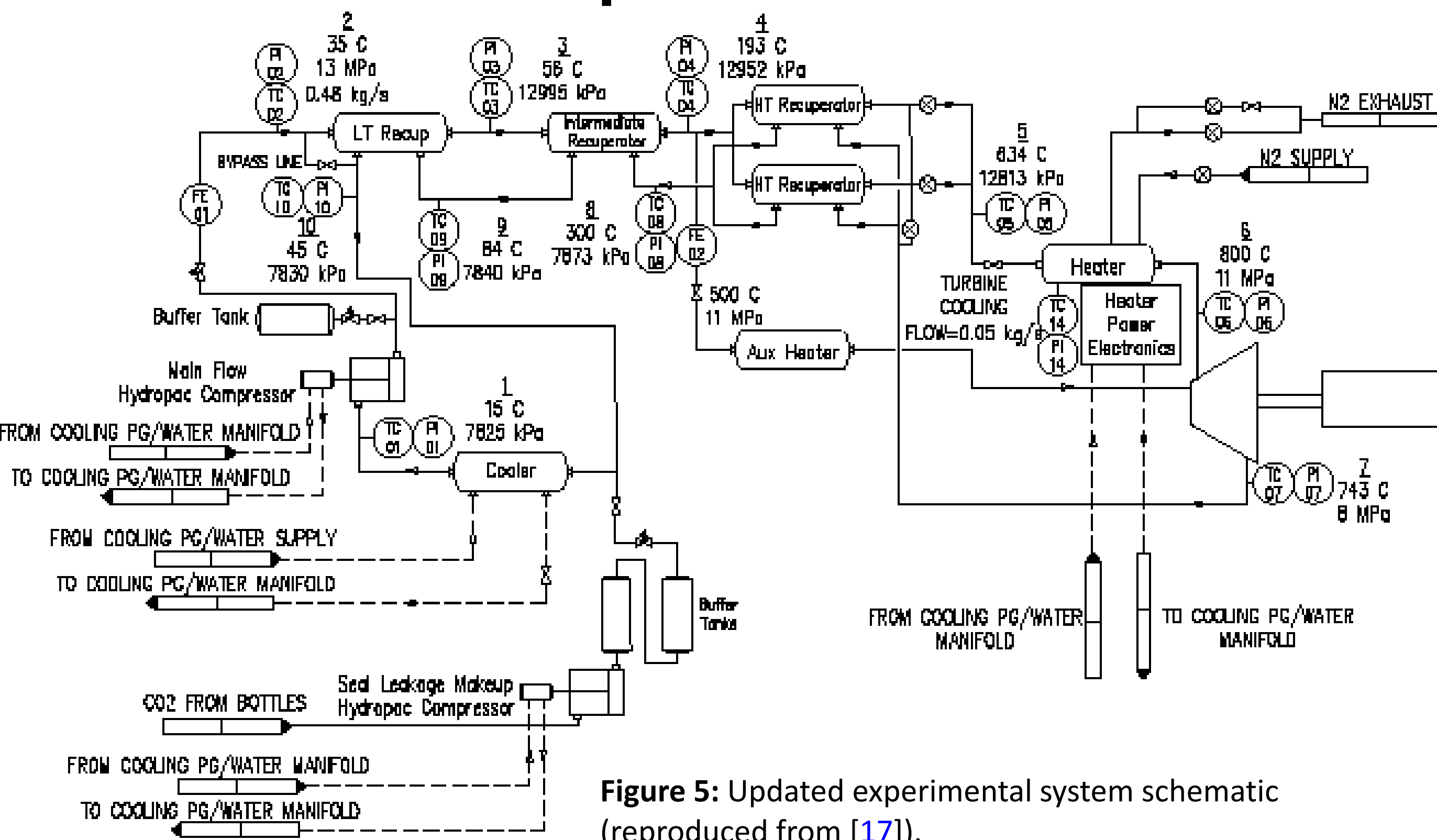


Figure 5: Updated experimental system schematic (reproduced from [17]).

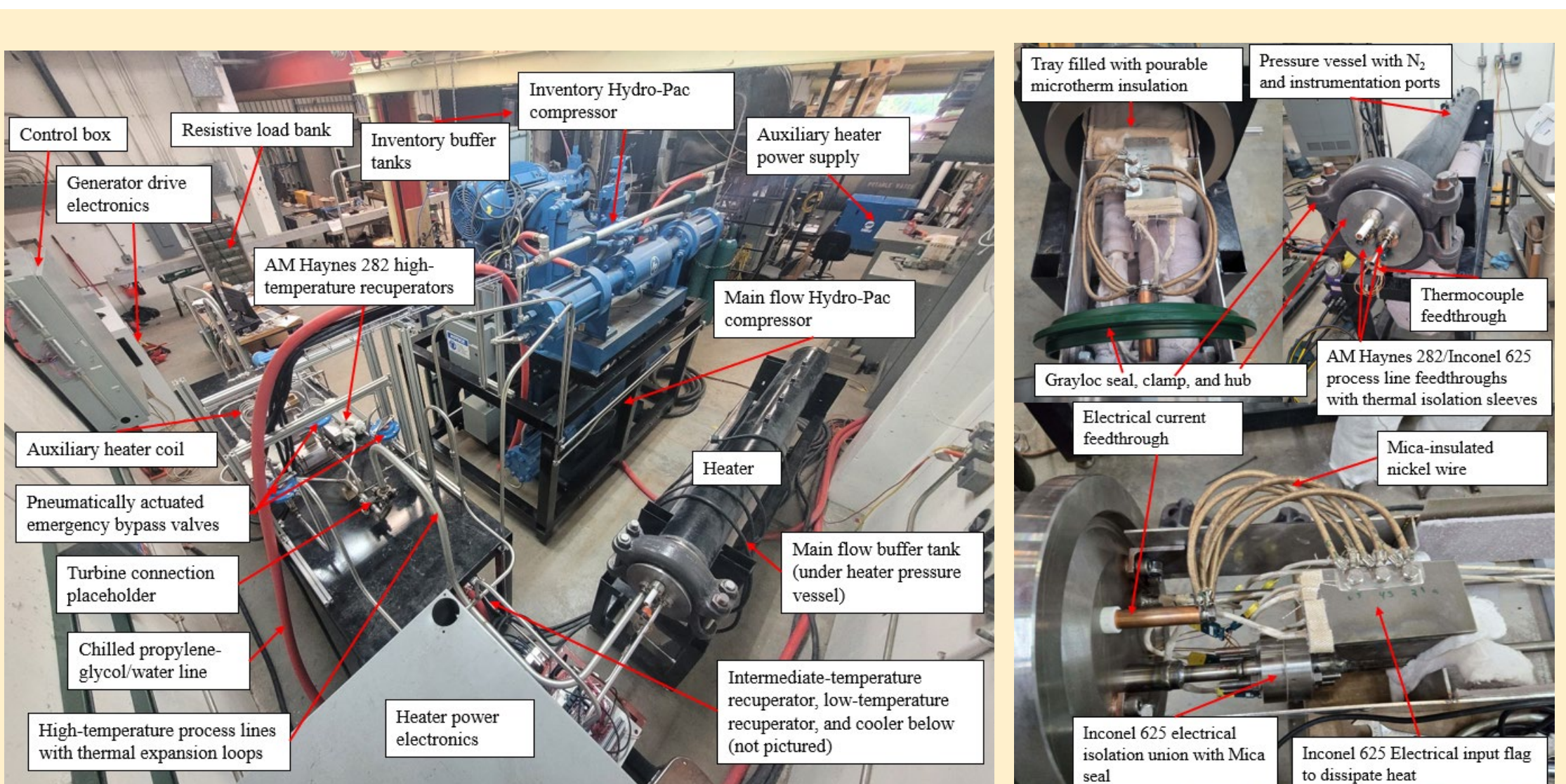


Figure 6: Constructed sCO₂ turbine test facility (reproduced from [17]).

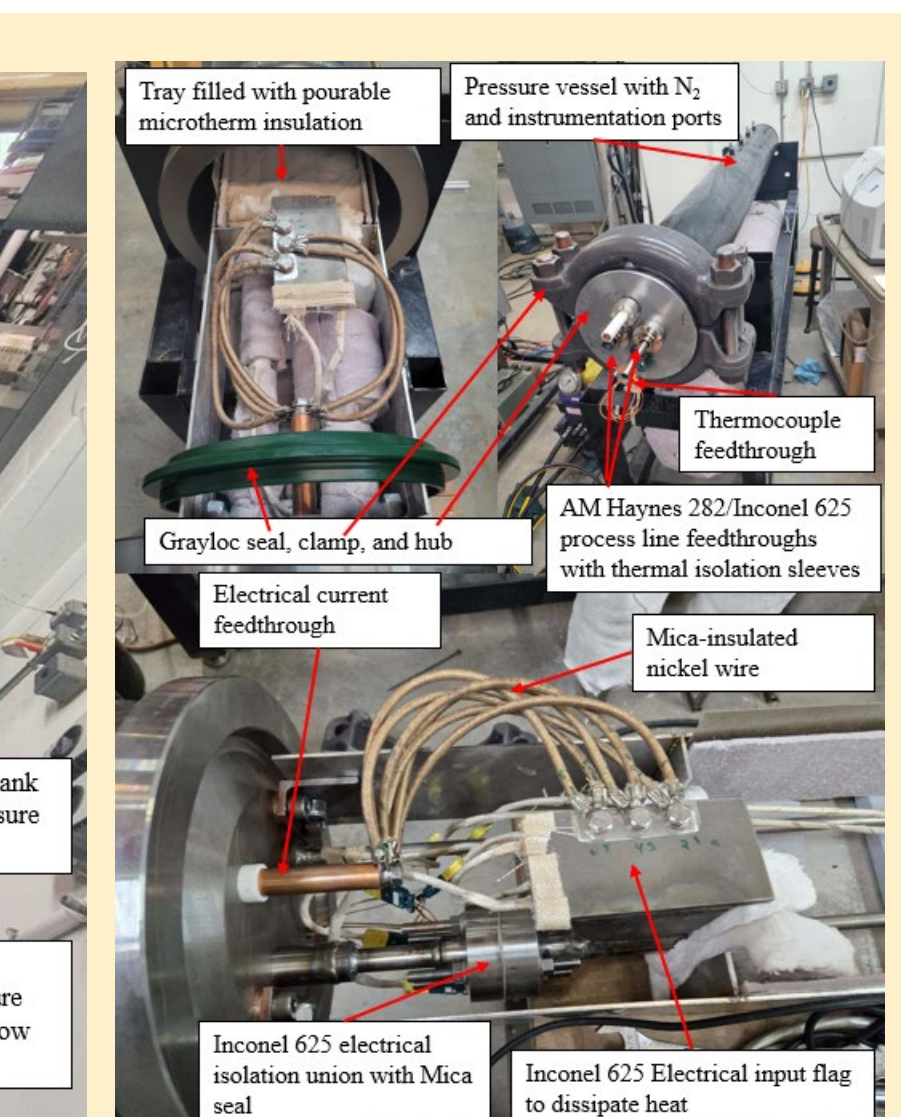


Figure 7: Constructed sCO₂ heater assembly (reproduced from [17]).

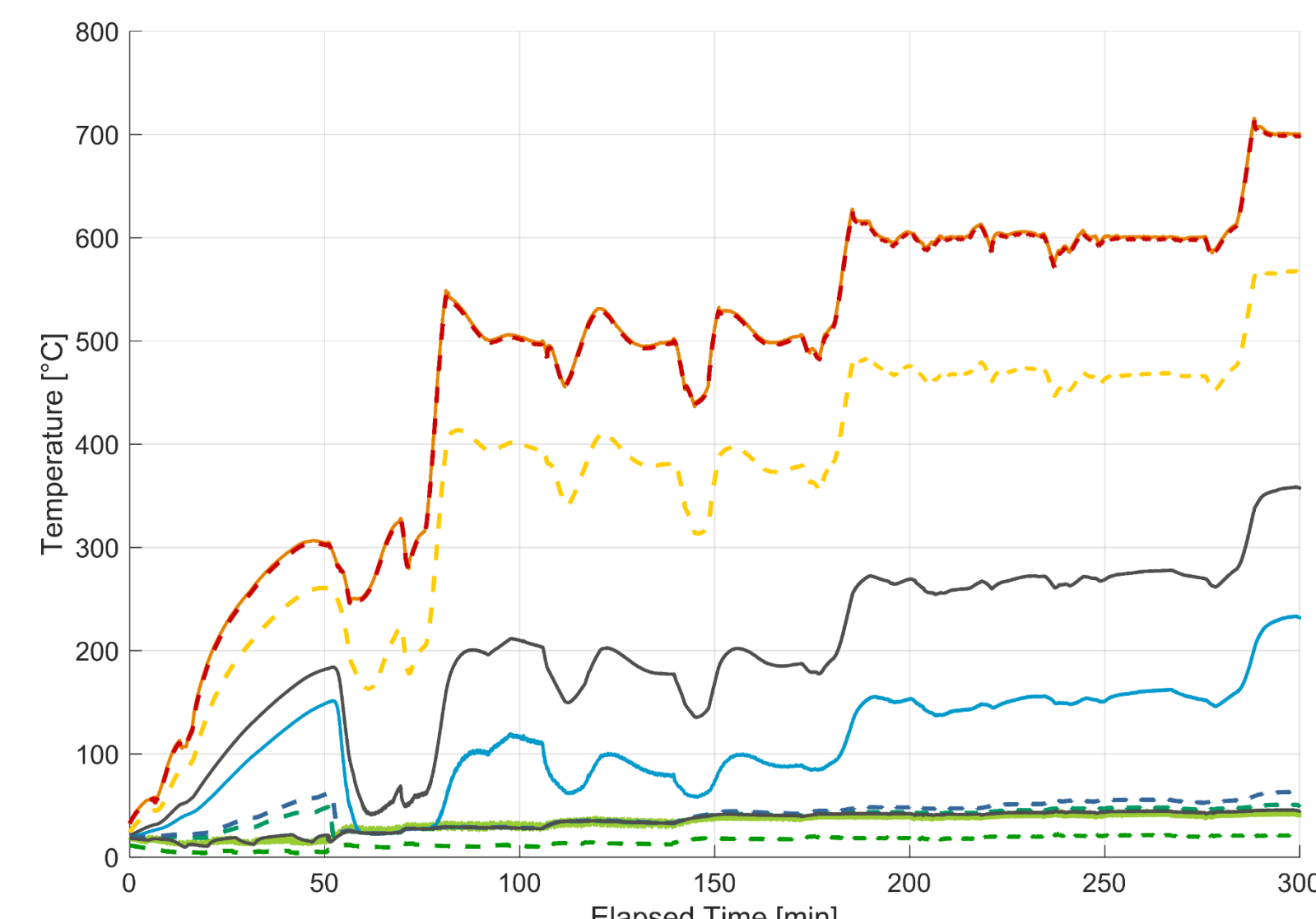


Figure 9: Measured temperature vs. time for all ten thermodynamic states during the validation campaign.

Integrated loss models dominate at smaller power scales.

Conventional optimal operating conditions are still valid at large power scales, but simple, subcritical cycles are optimal at small power scales.

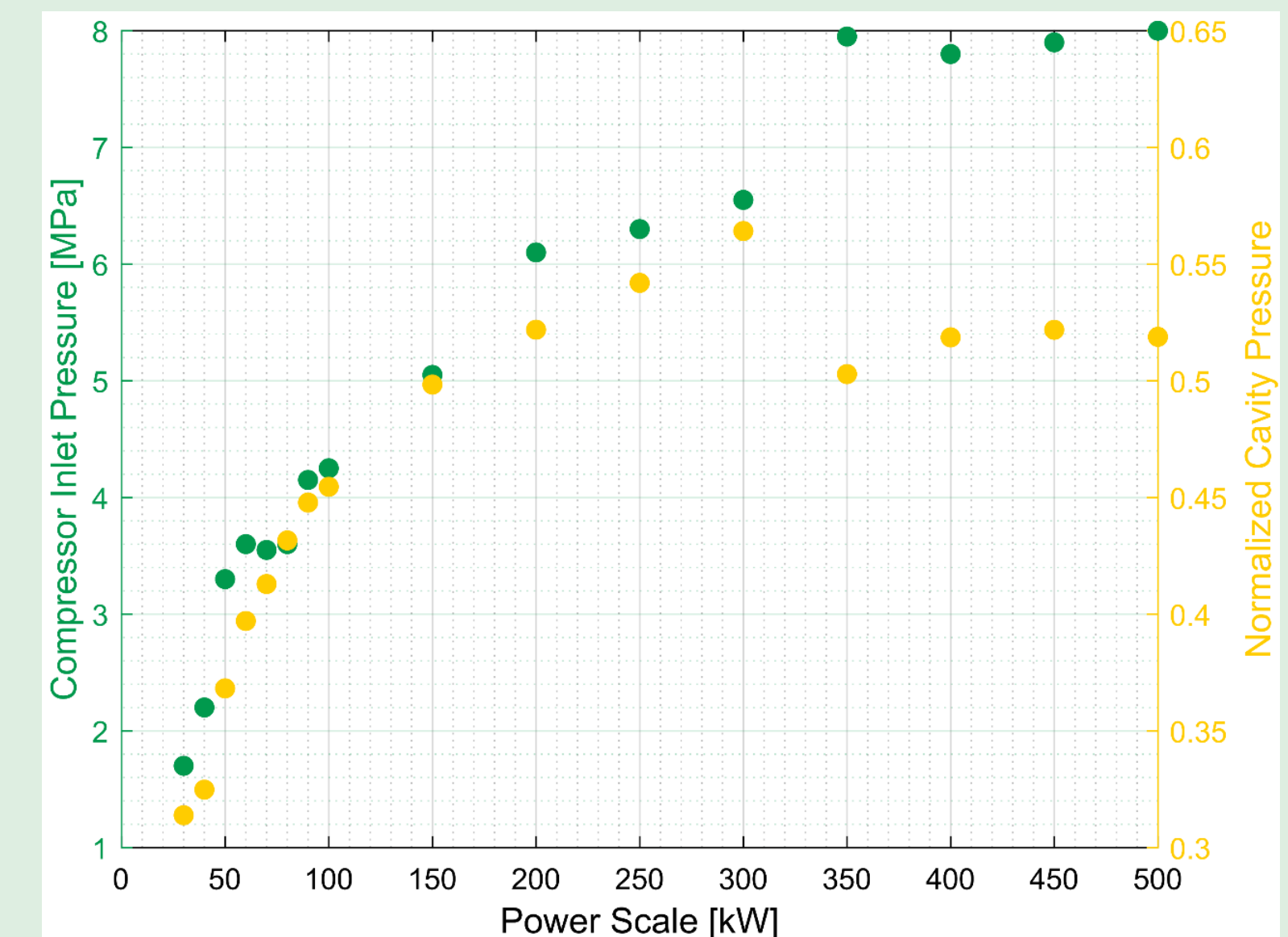


Figure 4: Optimal compressor inlet pressure and normalized cavity pressures as a function of power scale, derived from Pareto-optimal designs.