



Study of the effect of CO addition in a Direct Fired Oxy-Fuel Combustor for sCO₂ Power Cycles using Direct Detailed Chemistry and Adaptive Mesh Refinement

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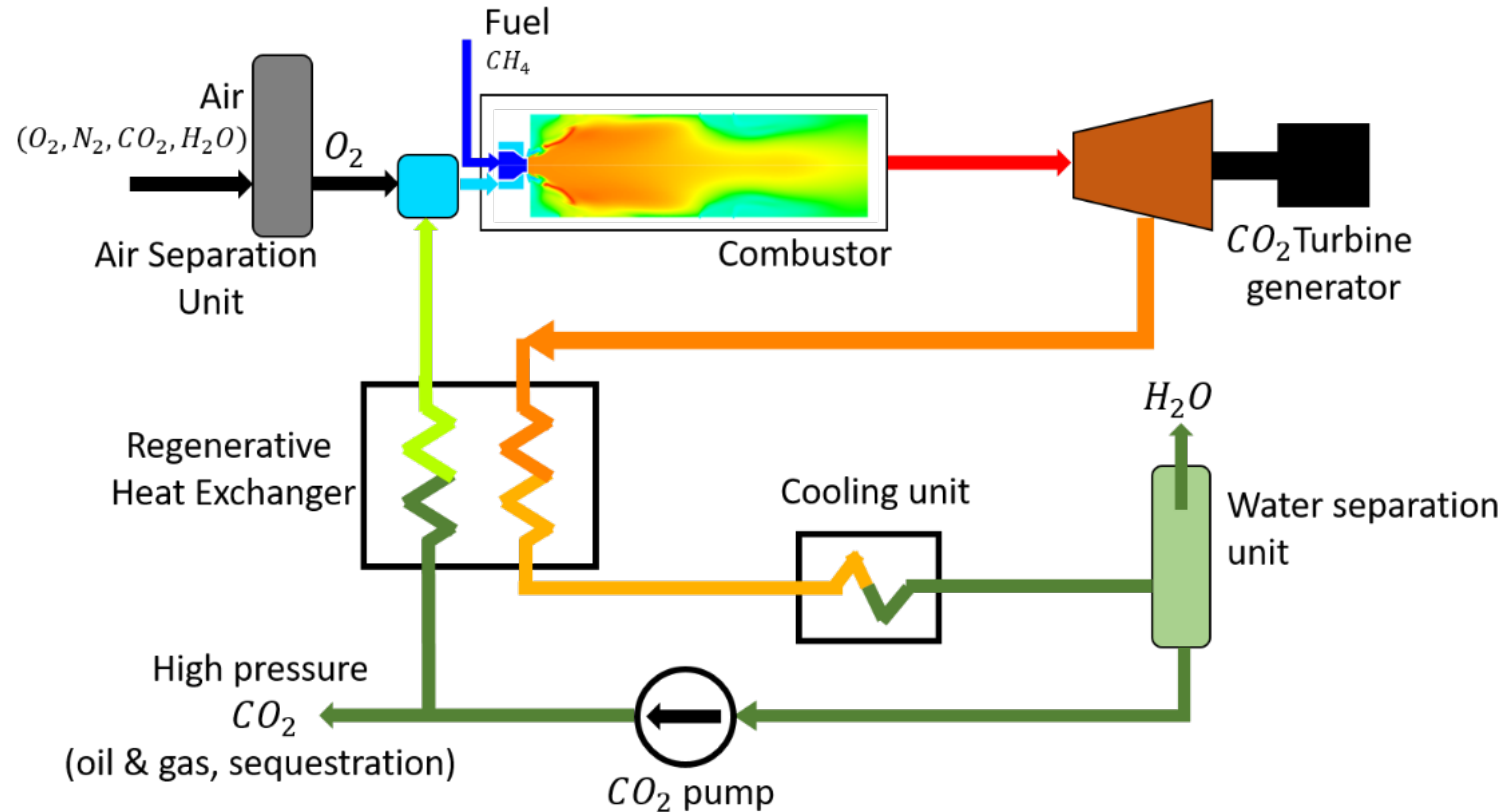
Introduction

- Supercritical CO_2 (sCO_2) power cycle is an emerging technology which has the potential to address both environmental concerns and energy demands
- Operating pressures at supercritical condition are in 200-300atm range
- At these extreme pressure conditions, experiments are expensive. Therefore, CFD modeling would play an ever-important role
- Managing impurities in the cycle is another foreseen stumbling block for successful operation
- Vasely *et al.* [1] showed that, impurities could significantly influence sCO_2 cycle performance
- Hence, it is **very crucial to understand the effect of impurities on sCO_2 combustion**
- Sources of impurities in sCO_2 combustion:
 - Impurities in fuel, inefficiency of air-separation unit before combustor, inefficiency of water separation unit after heat exchanger.
 - Ineffective air-separation unit may not filter Ar and N_2 entirely.
 - **Water separation unit may not separate CO , H_2O and other minor combustion products which are coming from the exhaust stream.**

Goal of the Study

- sCO_2 combustors work in a semi-closed loop
- Exhaust CO_2 is reintroduced in combustion chamber after removing water and other impurities
- Not all CO is removed and a significant part of it can make its way back into the combustion chamber
 - Pathways: mainstream flow, effusion and dilution flow
- Closed sCO_2 loop can become unstable if a positive feedback is established
 - A small amount of CO in inflow stream(s), increases CO at outflow multiple fold
- **Goal: study the effect of CO addition (impurity)**
- **We investigate this problem using two approaches**
 1. *Simplified model*: perfectly stirred reactor
 2. *Full 3D CFD modeling* of the combustor

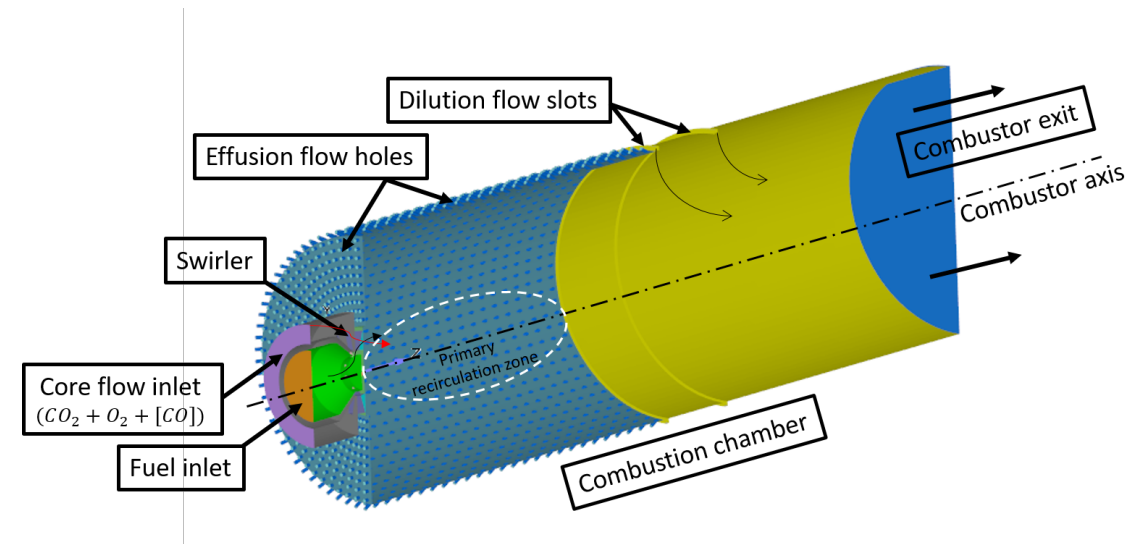
Supercritical CO_2 cycle



Schematic of Allam cycle which makes the basis for sCO_2 combustors

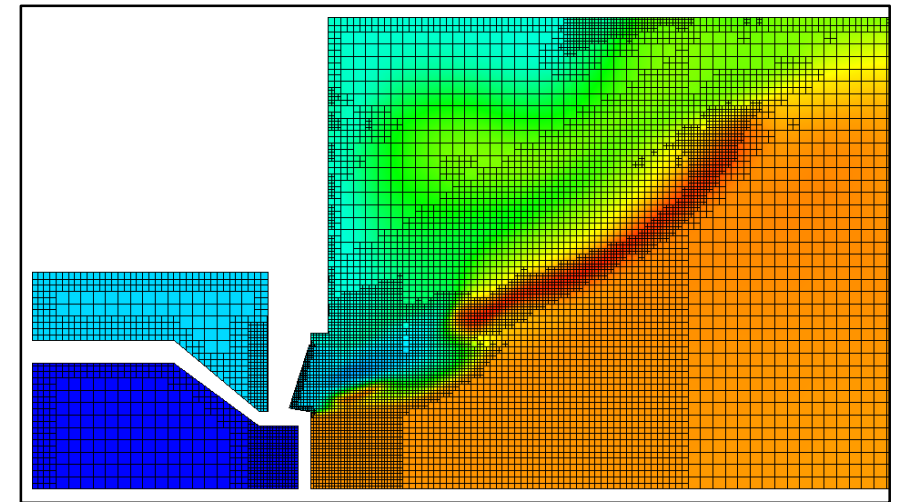
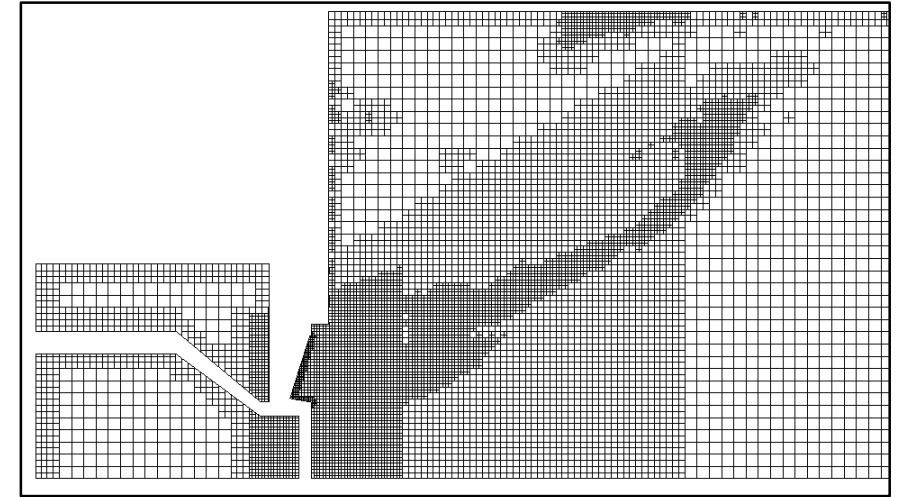
Geometry

- Concept oxy-fuel combustor geometry created at SWRI (Ref: Jacob Delimont *et al.*)
- Part of design study by SWRI, Thar Energy *et al.* for creating a 1MW (thermal) sCO_2 combustor
- Experimental in nature. Similarity with traditional gas fueled single axial combustors
- Simplified in-order to facilitate a parametric design study
- Combustor zones: swirler, primary zone and dilution
- CO_2 captured from exit cycled back in combustor through core inflow, effusion holes and dilution slots.
- Core flow composed of O_2 (obtained from air-separation unit upstream) premixed with super critical $CO_2 + CO$ impurities
- Fuel (CH_4) injected through circular holes along inner diameter in the swirler.



Numerical Modeling and Mesh

- Turbulence modeling
 - Realizable $k - \epsilon$ turbulence model
- Chemistry model
 - Cai-2017 (Cai, 2017)
 - Saudi ARAMCO 2.0 (W.K. Metcalfe, 2013)
 - No Adaptive Zone used
- Mesh settings
 - Base mesh: 2mm
 - Fixed refinement on walls
 - $Y^+ \sim 20$
 - AMR max refinement level = 3
 - Smallest cell size = 0.25mm
 - Cell count $4M$ cells



Adaptive mesh refinement for temperature in the recirculation zone

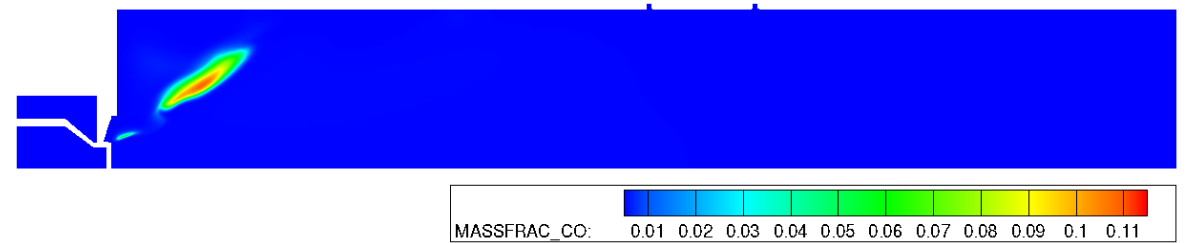
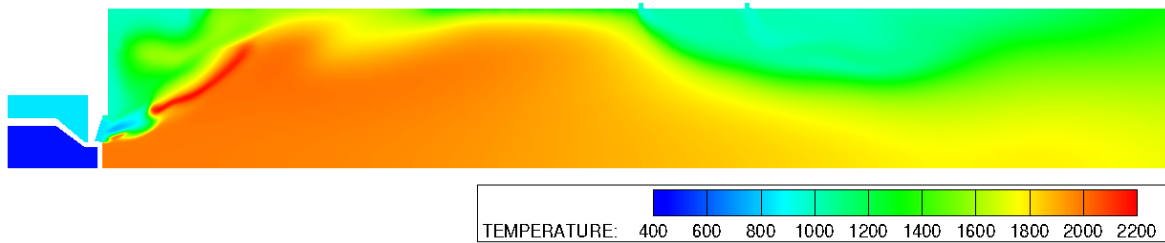
Effect of Mechanism

We study two mechanisms widely used in the CFD community for CH_4 combustion

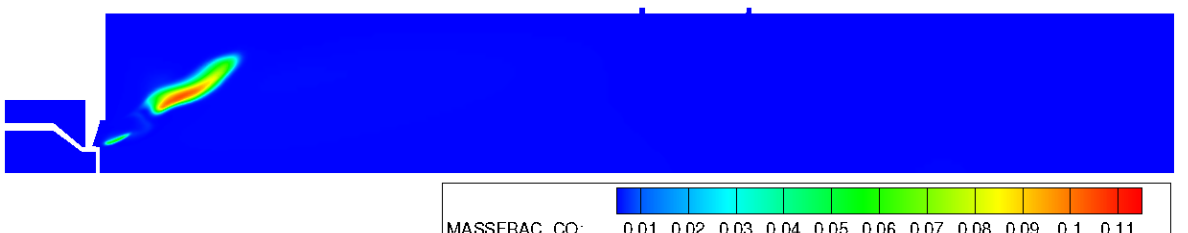
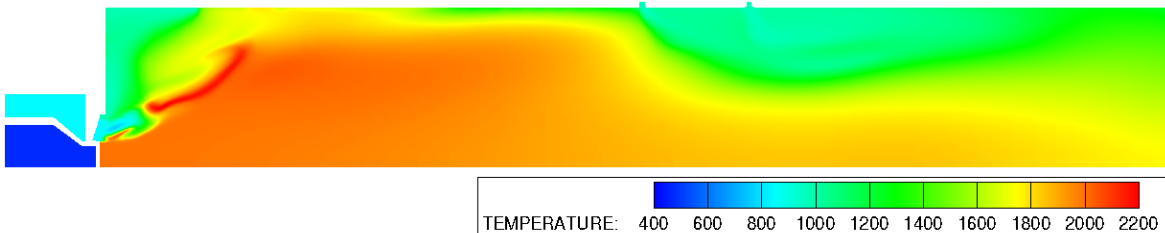
- Cai-2017 (Cai, 2017)
 - Developed at RWTH Aachen (Germany)
 - Suitable for oxy fuel combustion at high pressure (~30bar)
- Saudi ARAMCO 2.0 (W.K. Metcalfe, 2013)
 - AramcoMech 2.0 builds upon AramcoMech1.3
 - Developed by Combustion Chemistry Centre (C^3) at NUI Galway (funded by Saudi Aramco)
 - <https://www.nuigalway.ie/combustionchemistrycentre/#>
 - Reduced version of this mechanism (73 species) has been used in this work
 - Developed to characterize kinetic and thermochemical properties of large number of $C_1 - C_4$ based hydrocarbon and oxygenated fuels
 - Validated at very high pressures

Effect of Mechanism

- Goal
 - Qualitatively show difference between two mechanisms for methane combustion
 - Compare prediction of flow field, flame shape and emissions in the concept sCO_2 combustor
 - For this part, no CO is added to inflow stream
- Conclusion
 - Similar temperature and CO profiles. Some differences in flame shape and temperature in corner recirculation
 - Overall, both mechanisms perform equally well for sCO_2 combustion.



Temperature (left), CO (right) distribution from Saudi ARAMCO 2.0

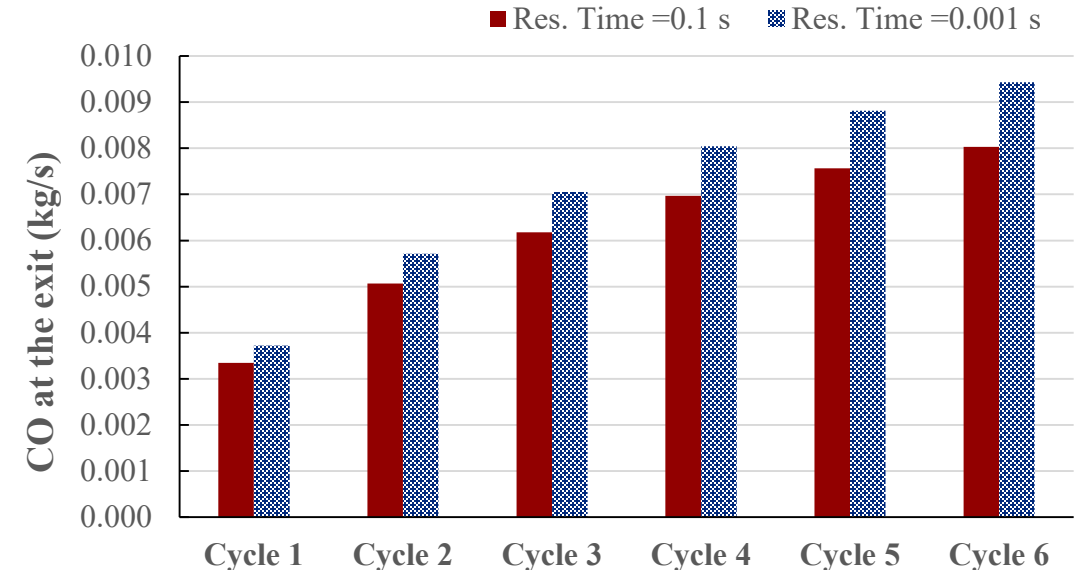


Temperature (left), CO (right) distribution from Cai-2017

Effect of CO Addition: PSR Modeling

- Perfectly Stirred Reactor (PSR) inflow conditions
- We first study effect of residence time on exit CO
 - Case has 75% CO_2 dilution
 - Two residence times studied: 0.001s and 0.1s
 - 0.1s t_{res} is representative of typical combustor
 - CO from cycle $N - 1$ is introduced in N^{th} cycle
- Growth of CO is faster in low t_{res} PSR
- Expected as expected: in low t_{res} PSR, time not sufficient to oxidize CH_4 to CO_2
- Remainder of the study: t_{res} PSR = 0.1s

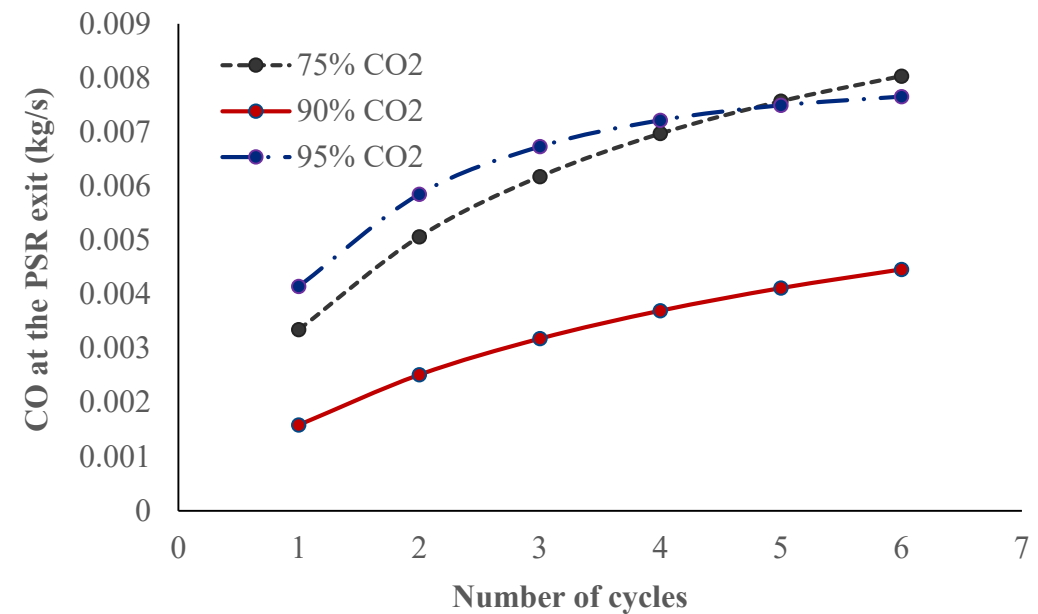
PSR inlet species	Flow rate
CH_4	0.02 kg/s
O_2	0.08 kg/s
CO_2	75%, 90% and 95%



CO at PSR exit for different residence time cases

Effect of CO Addition on exit CO : PSR Modeling

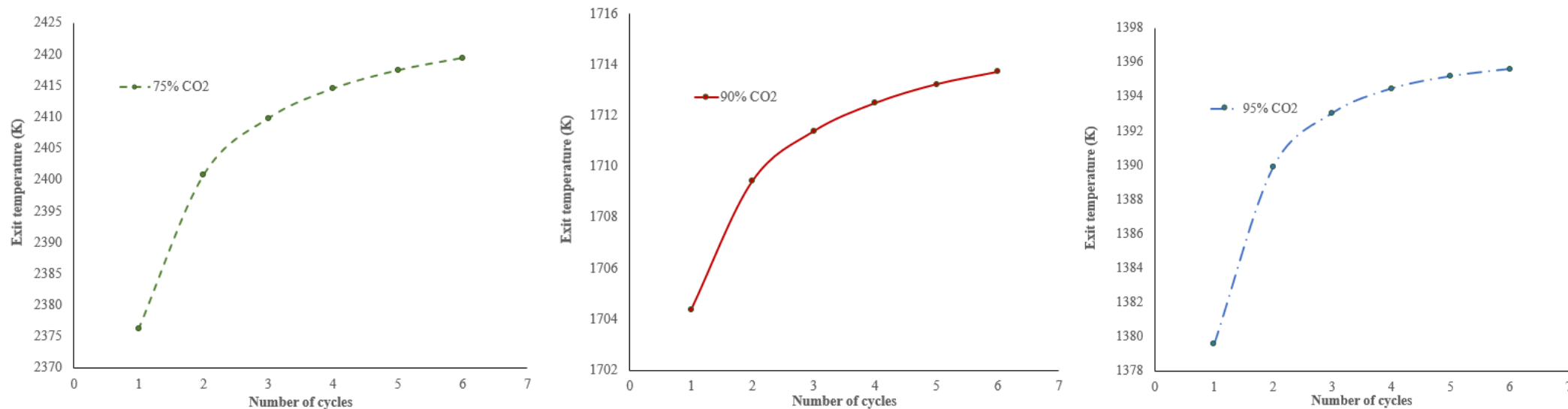
- sCO_2 systems work in a semi closed loop \rightarrow exit CO can come back into combustor through recycled CO_2
 - Potential to make sCO_2 system unstable
- PSR modeling
 - N^{th} PSR cycle, uses exit CO from $(N - 1)^{th}$ cycle. Cycle 1 has no CO .
 - Exit CO increases in consecutive cycles for all three CO_2 dilution mass-fractions
 - Trend it not exponential, as suspected by some in the sCO_2 community
 - Exit CO appears to settle down to a steady value in couple of cycles



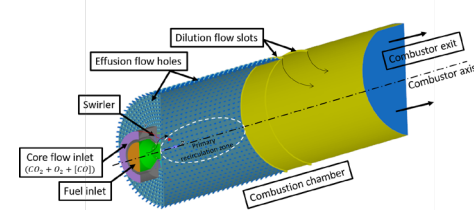
Evolution of exit CO in the PSR for different CO_2 dilution cases. Each cycle uses the exit CO of the previous cycle as inflow condition for CO . Cycle 1 has no CO at inflow.

Effect of CO Addition on T_4 (exit temp): PSR Modeling

- Evolution of exit temperature for different CO_2 dilution levels
- As CO_2 dilution increases \uparrow from 75% \rightarrow 95%, exit temperature decreases \downarrow over all
- For each case, exit temperature shows a trend to reach a steady value which is in-line with the trend in CO as oxidation of CO is the major contributor to overall heat release.



Evolution of exit temperature with cycles in different CO_2 dilution cases.



Effect of CO Addition on exit CO : 3D CFD Model

- Full 3D CFD using steady RANS and direct detailed chemistry used to simulate SWRI concept combustor
- If measured mass/mole fraction of CO at outflow $>$ CO introduced into the combustor \rightarrow indicator of a +ve feedback
- Δ increase in CO at exit is **less** compared to what is added to the inflow
- Above argument could be misleading
 - **Why?** Amount of CO_2 at inflow and outflow are different
 - **Correction:** Look at the ratio of mole fraction of CO to CO_2 at the inflow and outflow

	Outflow CO (kg/s)	
	ARAMCO 2.0	Cai-2017
Inflow $CO = 0$ kg/s	1.6×10^{-5} kg/s	1.6×10^{-6} kg/s
Inflow $CO = 4.67 \times 10^{-5}$ kg/s	4.1×10^{-5} kg/s	4.0×10^{-5} kg/s
Inflow $CO = 6.1 \times 10^{-5}$ kg/s	4.5×10^{-5} kg/s	4.6×10^{-5} kg/s

Mass flux of CO (kg/s) at combustor outflow

	Outflow $X_{frac}(CO)/X_{frac}(CO_2)$	
	ARAMCO 2.0	Cai-2017
Inflow $X(CO)/X(CO_2) = 0$	7.6×10^{-5}	6.3×10^{-6}
Inflow $X(CO)/X(CO_2) = 1.9 \times 10^{-4}$	1.5×10^{-4}	1.6×10^{-4}
Inflow $X(CO)/X(CO_2) = 2.5 \times 10^{-4}$	1.9×10^{-4}	1.93×10^{-4}

Ratio of mole fractions of CO to CO_2 at outflow

Summary

- Numerical framework
 - Perfectly Stirred Reactor (PSR); Full 3D RANS simulation with direct detailed chemistry and Adaptive Mesh Refinement for capturing flame shape and flow gradients
- Two key mechanisms (ARAMCO 2.0 and Cai-2017) studied for CH_4 combustion
 - Predicted temperature and CO profile very similar
 - Exit CO prediction also very close
- Investigated effect of CO addition
 - Determine if a concept sCO_2 combustor (designed at SWRI) establishes a +ve feedback loop, which would adversely affect the performance of the combustion system
- CO addition in the outflow does not seem to lead the sCO_2 combustor (working in a semi closed loop) in a positive feedback loop
- Exit CO tends to reach an equilibrium value (PSR model) or reduce (3D CFD) compared to what is introduced at the inflow end



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