

Symposium

sCO₂ Cooling Performance in Turbomachinery **Grant Musgrove (Southwest Research Institute[®], SwRI)**

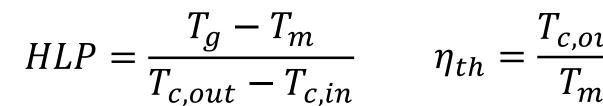
ABSTRACT

Current designs for sCO₂ power cycles use indirect heating to achieve turbine inlet temperatures of 775°C, which requires the use of advanced nickel alloys in the turbine. Because cycle efficiency is closely tied to the turbine inlet temperature, higher turbine inlet temperatures will be needed in the near future. In fact, temperatures as high as 1,200°C are already being achieved using oxy-combustion technology that is expected to be used in sCO₂ power cycles. To withstand higher inlet temperatures, the simplest solution is to integrate internal cooling into the sCO₂ turbine similar to conventional gas turbines in power generation and propulsion applications. Because the cooling fluid would be sCO₂ instead of air, the real gas effects must be included in the cooling design. In this paper, the real gas effects of sCO₂ are discussed using a cooling effectiveness chart, which is commonly employed to evaluate cooling designs in air-cooled gas turbines.

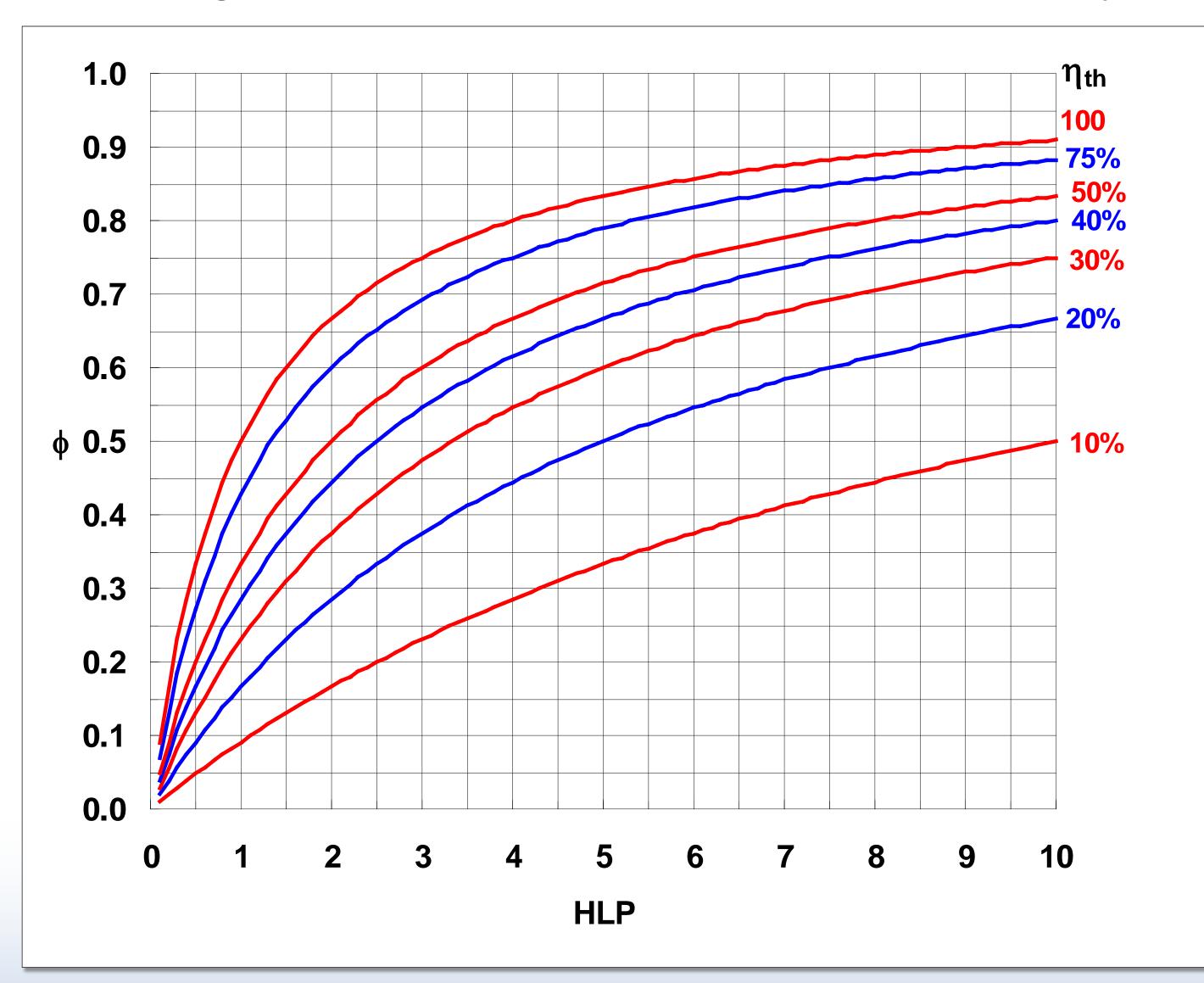
CONCLUSIONS

- The high specific heat of CO2 near the critical point increases turbine cooling effectiveness
- sCO₂ cooling designs could utilize a closed-circuit to recuperate heat gained by the coolant
- The pressure of the coolant flow can be modified to affect the overall the cooling performance

$$\phi = \frac{T_g - T_m}{T_g - T_{c,in}}$$



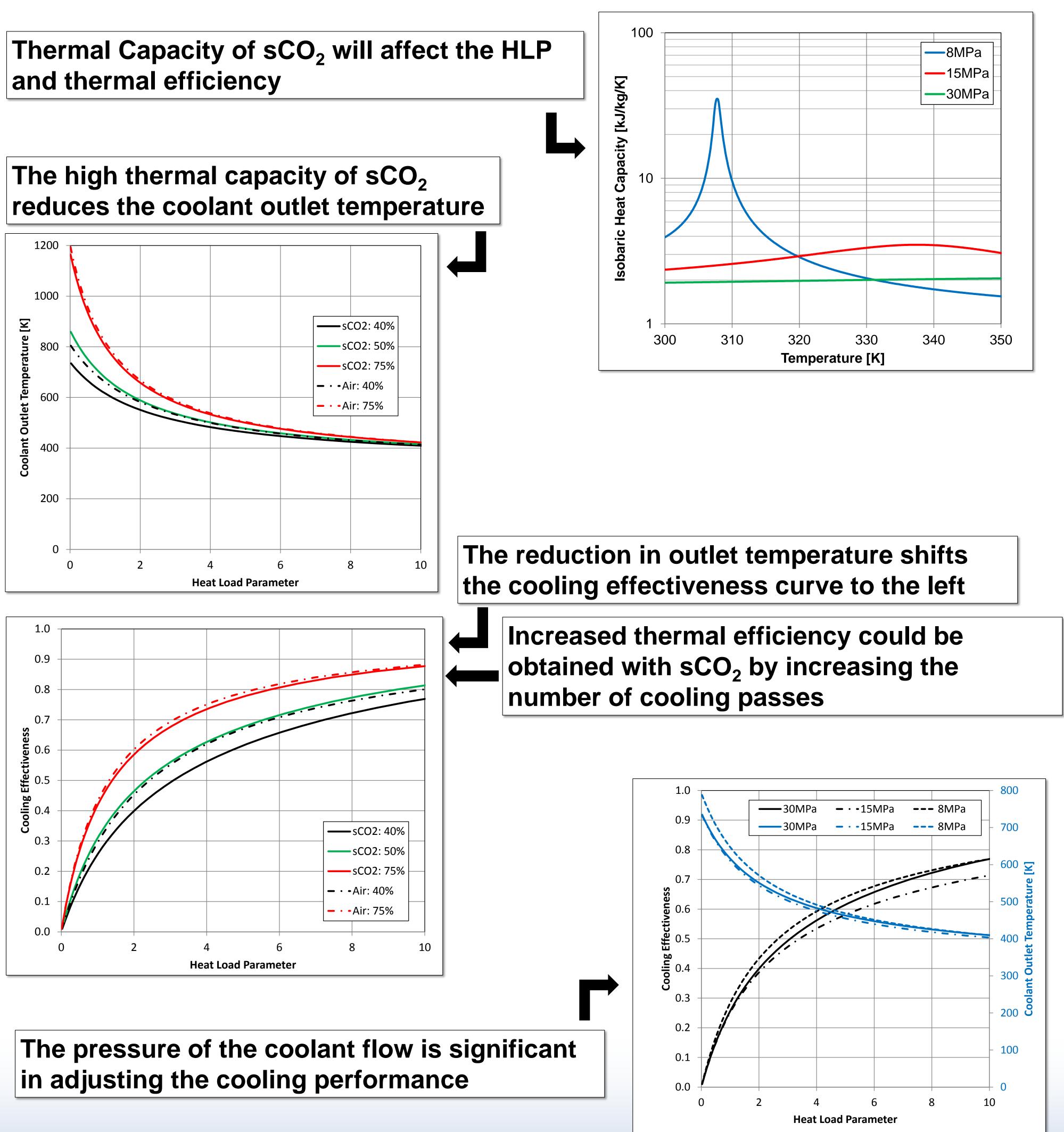
Cooling effectiveness Heat Load Parameter Thermal Efficiency



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Real gas effects of sCO_2 provide additional design variables to improve cooling performance

$$\frac{T_{c,out} - T_{c,in}}{T_m - T_{c,in}}$$





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