## Turbo-Machinery Considerations Using Super-Critical Carbon Dioxide Working Fluid for a Closed Brayton Cycle

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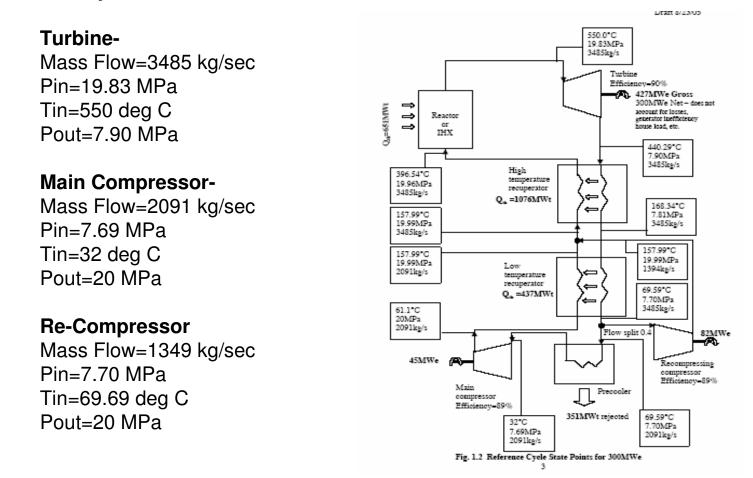
# **Machinery Discussion**

- Large Scale ~300 Mwe Rotating Equipment Design
  - Compressors
  - Turbine
  - Seal Options
  - Bearing Options
  - Generator Option
- Small Scale ~ 280 kWe System Design
  - Compressors
  - Turbine
  - Seal Options
  - Bearing Options
  - Generator Option



### Cycle Design Is the Input to the Turbomachinery Design

MIT Cycle (V. Dostal, M.J. Driscoll, P. Hejzlar, and N.E. Todreas, 2002 (MIT-ANP-TR-090)



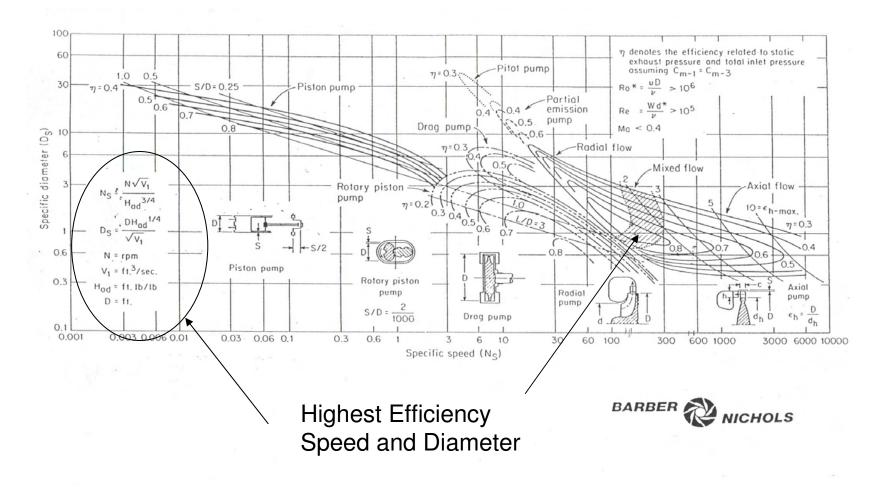


# Ground Rule: Industry Acceptance Is Important

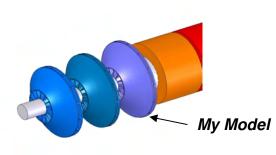
- Make the Machinery Look Like Conventional Power Plants If Possible (Generate Industry Interest)
  - 3600 RPM (GE Makes a 3600 RPM Hydrogen Cooled Generator @ ~300 Mwe)
  - Single Shaft
  - Oil Lubricated Hydrodynamic Bearings (Tilt Pad or Elliptical)
  - Seals (Acceptable on Steam and Gas Turbines)
  - Horizontal Shaft



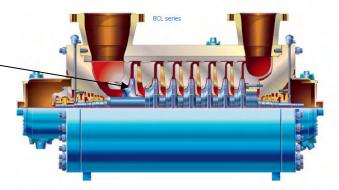
### Ns-Ds Diagram Compressors (English Units)







From General Electric Site BCL Series Compressor

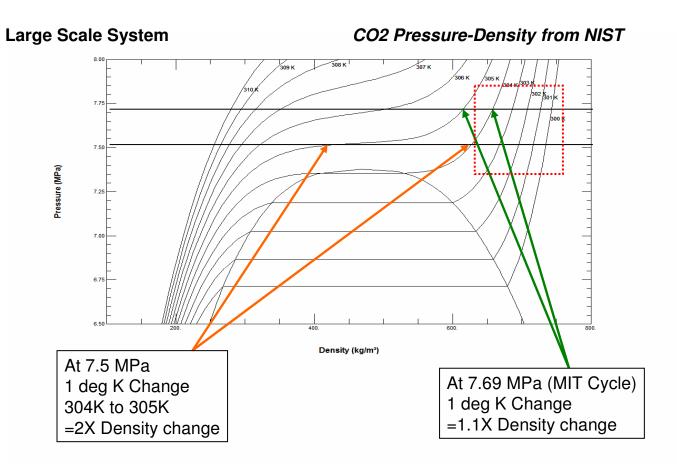


#### Main Compressor (Large System) 3-Stage Radial Efficiency is 87%

MAIN CO THREE 3600 RP		ERFORMA RADIAL Stage eff	NCE 0.87	Wdot 2091kg/s Mt 300MW cycle *Stage enthalpies need to be corrected but are OK for concept study					
Temperatu (°C)	ure Pressure (MPa)	Density (kg/m³)	Enthalpy (kJ/kg)	Entropy (kJ/kg-K)	Op (kJ/kg-K)	Stage Ns	US Ns	Efficiency Overall	Stage U2 ft/s ia inches
32 41.649	7.69 10.766	598.81 647.38	306.81 311.74	1.3483 1.3483	15.813 4.7145	1.20	154		343.53 → 21.87
41.803 51.164		644.8 684.11	312.48 318.53	1.3506 1.3506	4.7558 3.1544	0.99	128		361.03 → 22.98
51.447	14.803	681.47	319.43	1.3534	3.1672	0.82	106		371.98
60.965 61.405	20.014	717.81 715.02	326.88 327.99	1.3534 1.3567	2.5125 2.5179	$\backslash$	*	0.07	23.68
60.286	20.014	722.1	325.18	1.3483	2.5041		*	0.87	
							$\checkmark$		
3-Stage Radial (Mixed Flow) Compressor 3-Stage Meets Target Efficiency Overall Efficiency									



#### Large Density Variation for Main Compressor Makes for a Difficult Design



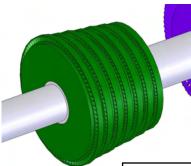
Need to Keep Compressor Inlet Density In Small Range for Successful Operation



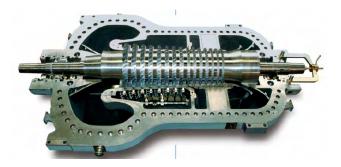
# Main Compressor Radial Type

- Close to the Dome During Startup/Shutdown/System Upset/Wet Gas Handling
- Radial Head/Flow Characteristics for Startup/Shutdown Flow/Pressure Transients
- Flat Head v Flow Characteristic Allows Maintenance of Head over a Wider Flow Range
- Reduced Number of Stage for Overhung Configuration (Rotordynamic Consideration)
- Shrouded Design for Best Efficiency





### Re-Compressor Axial Type



	Axial Re- Compressor 7-Stage						
	Stage						
	1	2	3	4	5	6	7
Pressure Ratio	1.226	1.2033	1.18	1.146	1.1126	1.0914	1.0711
P in psia	1116	1369	1647	1945	2228	2480	206
P out psia	1369	1647	1945	2228	2480	2706	2899
Specific Speed	144	138	134	141	157	171	194
D tip (inches)	42	40.7	39.5	38.3	37.1	35.8	34.6
D hub (inches)	39.1	37.8	36.7	35.5	34.4	33.2	32
Hub/Tip Ratio	0.930952	0.928747	0.929114	0.926893	0.927224	0.927374	0.924855

Hub/Tip Ratio is Above .9 (Needs Further Review)



Analysis Was Done For Multi-Stage Radial Compressors

Re-Compressor as Radial 84%

Difficult to Obtain High Efficiency

Three Stage	3070.5lb/sec									
Temperature	Pressure	Density	Enthalpy	Entropy	Ср		$\backslash$			
(°F)	(psia)	(lbm/ft <sup>3</sup> )	(Btu/lbm)	(Btu/lbm-°R)	(Btu/lbm-°R)					
157.26	1116.5	10.281	205.95	0.45305	0.38367	298.6577	6690.8 🔪	84.09706	453.73	
220.03	1675	13.848	214.55	0.45305	0.38935			$\backslash$	28.89	▼
223.98	1675	13.631	216.07	0.45528	0.3827	225.2586	5679.4	82.58789	396.58	
273.68	2275	16.834	223.37	0.45528	0.38298			$\backslash$	25.25	_ \
277.07	2275	16.641	224.66	0.45704	0.3789	184.5142	4986.98	82.40286	345.04	
317.58	2900	19.46	231.07	0.45704	0.37647			$\backslash$	21.97	
320.58	2900	19.289	232.2	0.45849	0.37382	159.184		×	0.84	$\sim$ V
309.46	2900	19.941	227.99	0.45305	0.38413					$\mathbf{N}$
										/λ



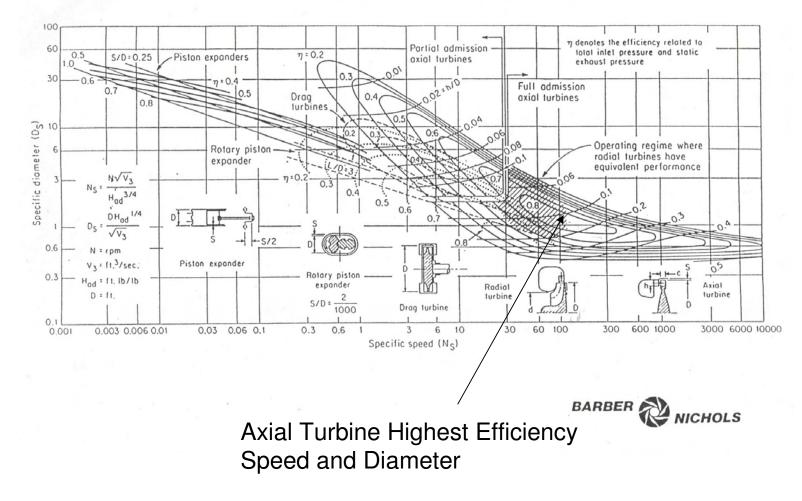
Large System

Wheel

Inches

Diameter /

### Ns-Ds Diagram Turbines (English Units)





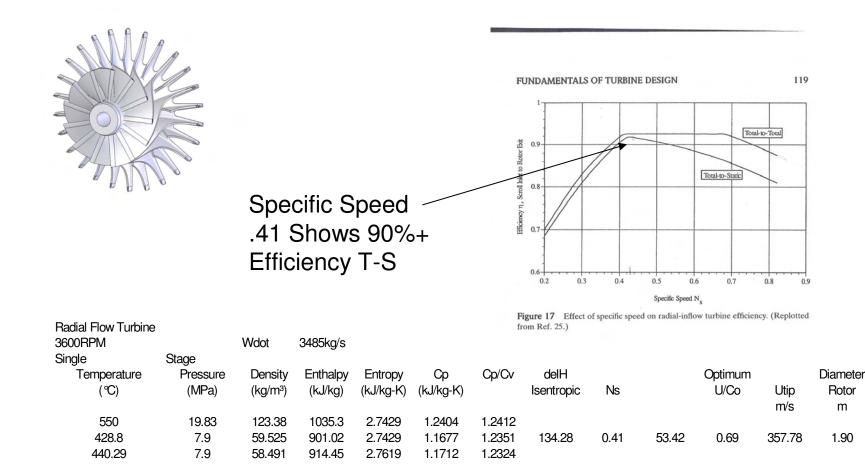
# Turbine Design (3-Stage Axial) 90% Efficiency



	Axial Reaction Turbine Summary						
3-stage							
	Stage 1	Stage 2	Stage 3				
Turb In Temp F	1022	955	888				
Nozz In Temp F	972	905	839				
Rotor Out Temp F	951	883	819				
Turb Out Temp F	955	888	823				
Mass Flow lb/sec	7683	7683	7683				
Adiab. Head B/#	19.735	19.735	18.59				
Hub Dia 1 Inch	28.59	36.43	32.14				
Hub Dia 2 Inch	32.23	38.64	35.2				
Tip Diameter Inch	45.687	47.58	48.9				
Reaction	0.4	0.4	0.4				
Blade Chord Inch	2	2	2				
# Blades	85	85	85				
Specific Speed	93	105	125				

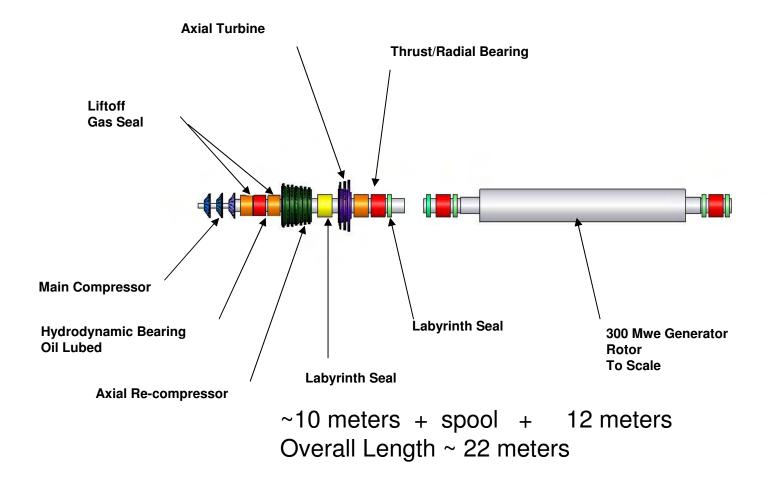


# Turbine (Single Stage Radial) 90% Efficiency ~1.9 meter Diameter





#### 300 Mwe Super-Critical CO2 Closed Brayton Cycle Rotating Group





# Large Scale System Oil Lubricated Hydrodynamic Bearings

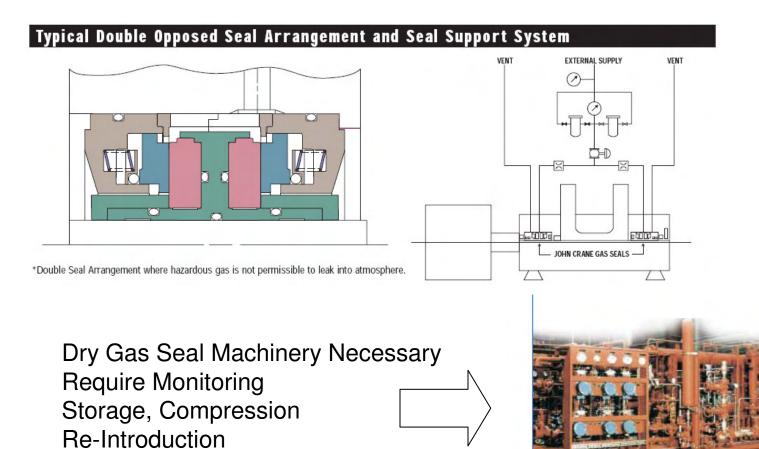
-Thrust and Journal Hydrodynamic Bearings (Industry Standard for Power Generation Equipment, Waukesha)







### Liftoff Gas Seal (John Crane) Surface Speeds/Pressures/Temperatures/CO2 Currently Offered





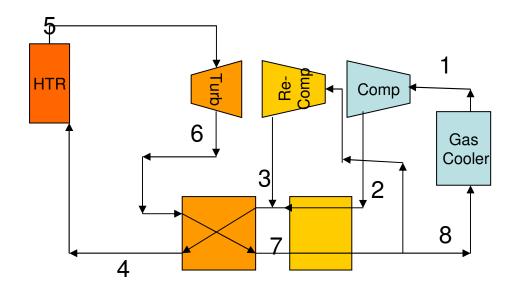
(From GE Site)

# Small Scale System ~300 kWe

- Study SCO2 Closed Brayton Cycle on Small/Affordable Scale
- Same Pressures and Much Lower Flow Rate
  - Higher Speed Machinery to Gain Efficiency
  - Radial Compressors and Turbine
  - High Speed PM Motor/Generator
  - Bearings/Seals for Large System Not Optimum for Small System



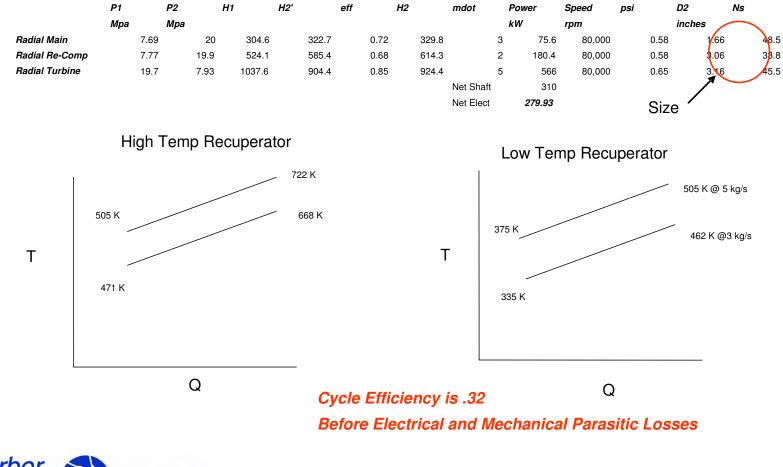
### Small Scale Loop (Mass Flow 5 kg/s)



Station	Station T(K)		P (Mpa)	mdot (kg/s) eff	•	dP/P	kJ/kg	
	1	305	7.69	3	72	0.01	304.6	
	2	335.2	20	3			329.8	
	3	485	19.9	2	68	0.005	614.3	
	4	668	19.8	5		0.005	843.9	
	5	825	19.7	5		0.005	1037.6	
	6	722	7.93	5	85		924.39	
	7	504.7	7.85	5		0.01	675.89	
	8	375	7.77	3		0.01	524.07	



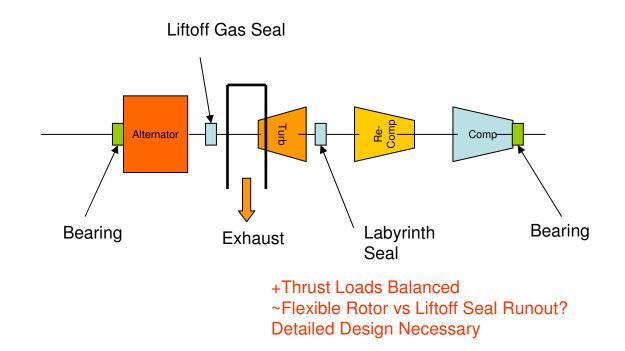
### Cycle Analysis with Pressure Drops ~279 kWe Net Electric Power



Barber Nichols

*Oil Lubricated Bearings with Seals to Use "Large Machine" Technology* 

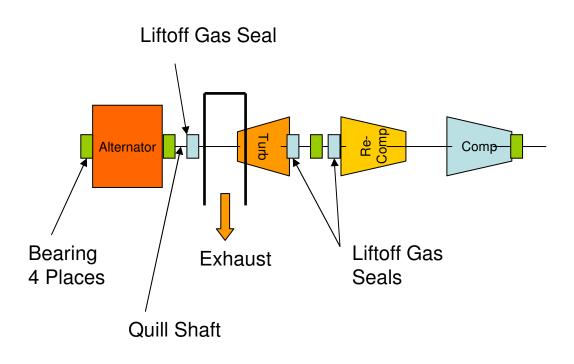
### Scaled Loop Machinery (2 bearing option)





*Oil Lubricated Bearings with Seals to Use "Large Machine" Technology* 

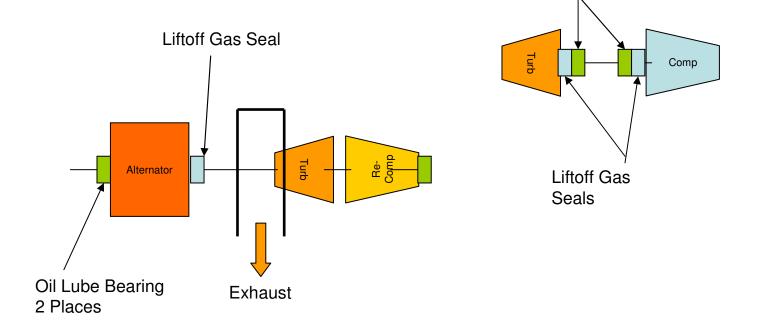
### Scaled Loop Machinery (4 bearing option, more seals)





*Oil Lubricated Bearings with Seals to Use "Large Machine" Technology* 

### Scaled Loop Machinery (2 Shaft Option)





# Simplified Design

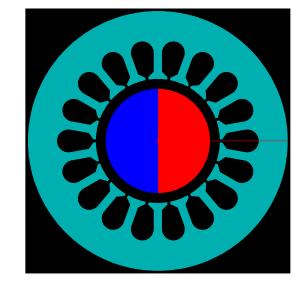
- CO2 Bearing Supply
  - Hydrostatic
  - Hydrodynamic
    - Flex Pad
    - Foil
- Generator Operating in CO2
  - Eliminate Gas Liftoff Seals/Laby Seals OK
  - Supercritical CO2 Degradation of Insulation
  - Windage Loss

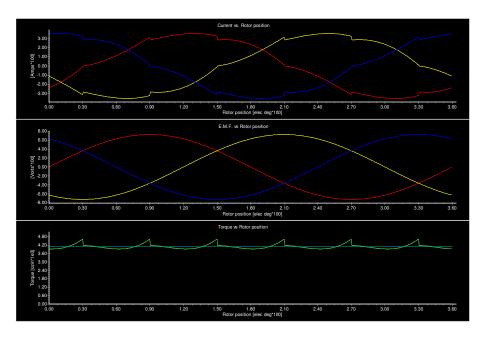


### Generator Technology Very High Power/Speed Compact for Rotordynamics

Permanent Magnet Generator -45 MGOe NIB Magnet -Arnon 5 Laminations -7" Stack Length -5" Outer Diameter -Inconel 718 Rotor Can -279 kWe Output at 80,000 rpm -98% Efficiency

-Windage in CO2 at 170 deg F -62 kW @ 1100 psi -11 kW @ 250 psi -1 kW @ 14.7 psi Need to Operate Generator at Low Pressure

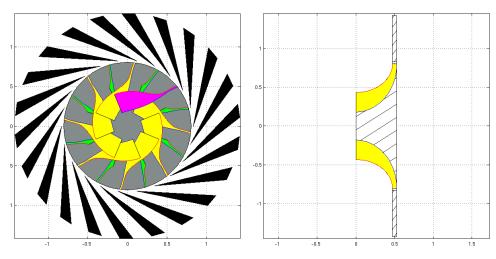






### Main Compressor Analysis

Looks More Like a Pump Than a Compressor





•Developed Defined Procedure

•Use Real Gas Mean Line Code

Modify for Ideal Gas

•Flow Path Analysis

•Sizing

•Input to CFD Code with Average CO2 Properties



## Other Considerations To Be Considered When Designing Turbomachinery

- Rotordynamics
- Thrust Load Management
- Startup/Shutdown Transients
- Clearances
- Inlet/Discharge Diffusion etc.
- Stresses (Including Thermal/Fatigue/Operating etc)

